

# POWER CONTROL UNIT



FLEXIBLE DESIGN • SCALABLE • RELIABLE



## OVERVIEW

Magellan's power control unit (PCU) serves as the interface between the spacecraft solar array, battery and power bus. The PCU manages power transfers from the solar arrays to the batteries and provides fault-protected switches between the batteries and other spacecraft subsystems and payloads. It also provides extensive telemetry for determining the health of the power bus, including battery A-hr integrators, as well as solar array and individual switch current monitors. Magellan's PCU is designed to be scalable, capable of meeting power needs from 100 W to 3 kW, and can be configured to be either single or dual string.

### EXAMPLES OF REPRESENTATIVE PCU CONFIGURATIONS

	Flagship-Class Mission <sup>1</sup>	Generic SmallSat Mission <sup>2</sup>
Card Complement:	<ul style="list-style-type: none"><li>• 2 controller cards (dual string)</li><li>• 4 switch cards with 24 switches each</li><li>• 2 propulsion/pyrotechnics cards</li><li>• 1 solar array card</li><li>• 1 battery card</li></ul>	<ul style="list-style-type: none"><li>• 1 controller card (single string)</li><li>• 1 switch card with 24 switches</li><li>• 1 solar array card</li><li>• 1 battery card</li></ul>
Power:	<ul style="list-style-type: none"><li>• Peak dissipation under peak load: &lt;50 W<ul style="list-style-type: none"><li>- Assuming 1,500 W peak spacecraft power, with 600 W from solar array and 900 W from battery</li></ul></li><li>• Dissipation under orbit-average load: &lt;35W<ul style="list-style-type: none"><li>- Assuming orbit-average spacecraft power demand of 500 W</li></ul></li></ul>	<ul style="list-style-type: none"><li>• Peak dissipation under peak load: &lt;35 W<ul style="list-style-type: none"><li>- Assuming 500 W peak spacecraft power, with 250 W from solar array and 250 W from battery</li></ul></li><li>• Dissipation under orbit-average load: &lt;30 W<ul style="list-style-type: none"><li>- Assuming orbit-average spacecraft power demand of 150 W</li></ul></li></ul>
Input Voltage:	<ul style="list-style-type: none"><li>• 28±6 V<sub>DC</sub> unregulated</li></ul>	<ul style="list-style-type: none"><li>• 28±6 V<sub>DC</sub> unregulated</li></ul>
Mass:	<ul style="list-style-type: none"><li>• 25 kg</li></ul>	<ul style="list-style-type: none"><li>• 12 kg</li></ul>
Dimensions (LxWxH):	<ul style="list-style-type: none"><li>• 32×37×23 cm</li></ul>	<ul style="list-style-type: none"><li>• 18×37×23 cm</li></ul>
Quiescent Power:	<ul style="list-style-type: none"><li>• 20 W</li></ul>	<ul style="list-style-type: none"><li>• 20 W</li></ul>

<sup>1</sup> Actual values for an upcoming flagship-class mission with an undisclosed customer.

<sup>2</sup> Representative values for a hypothetical small satellite mission.

### SCALABLE HARDWARE CONFIGURATION

The innovative “by the slice” modular design of the Magellan PCU allows customers to choose only the mission-specific functionality and level of redundancy required, thereby optimizing mass, power and volume. For example, a generic smallsat mission requiring less than 750 W could be supported with four cards (controller, solar array, battery, switch).

- Controller card
- Solar array card
- Battery card
- Switch card(s)
- Propulsion/pyrotechnics card

Enhanced functionality for more complex missions would require the inclusion of additional cards.

### RADIATION TOLERANCE

Total Ionizing Dose: >50 krad (Si)  
Single Event Latchup Threshold: 59 MeV·cm<sup>2</sup>/mg

### FLIGHT HERITAGE

RADARSAT Constellation Mission (launch in 2018)

## ABOUT MAGELLAN AEROSPACE

Magellan Aerospace (TSX: MAL) is a global company that provides products and solutions to the aviation, defense and space markets, with facilities in Canada, the United States, Europe and India. The company's Space and Rocket business has over fifty years of flight heritage on NASA and CSA missions including Black Brant sounding rockets, sub-orbital payloads, Shuttle and ISS experiments, and satellite missions. Magellan's space solutions include the microsatellite-class MAC-100 bus and the smallsat-class MAC-200 bus, as well as command and data handling (C&DH) and power control unit (PCU) subsystems.

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